

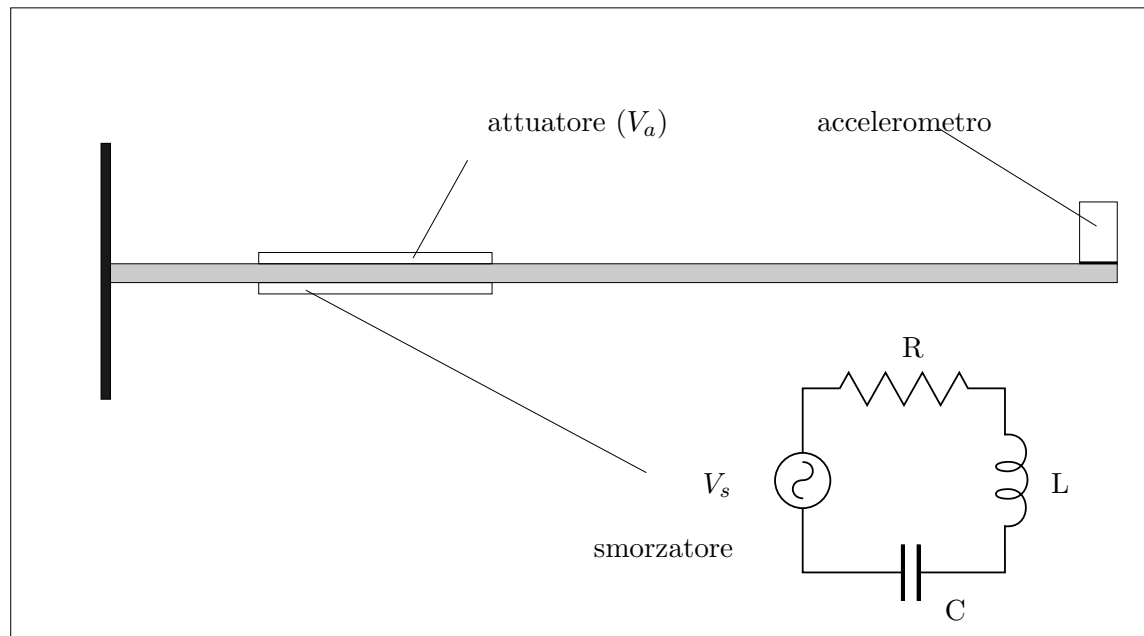
# SMORZAMENTO PASSIVO DI VIBRAZIONI ATTRAVERSO LAMINE PIEZOELETTICHE

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## Obiettivi

- riduzione dell'ampiezza delle oscillazioni dissipando energia meccanica attraverso un circuito elettrico
- realizzazione di un circuito risonante RLC con elevati valori di induttanza (richiesti a basse frequenze)
- simulazione di un'induttanza con un *giratore*
- confronto della risposta sperimentale con la risposta calcolata



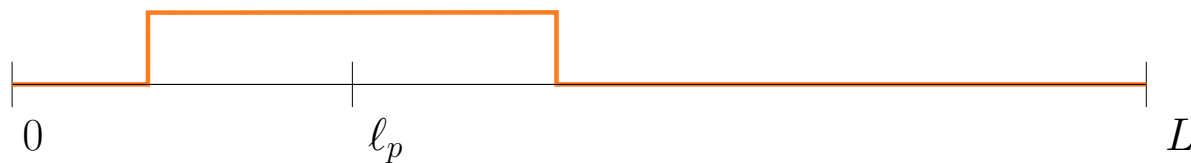
## Modello meccanico

$$W_{out}(w, \omega) = \int_0^L (b \cdot w + c \cdot \omega) d\xi + \mathbf{t}_0 \cdot w(0) + \mathbf{t}_L \cdot w(L) + \mathbf{m}_0 \cdot \omega(0) + \mathbf{m}_L \cdot \omega(L),$$

$$W_b(w, \omega) = - \int_0^L (z \cdot w + t \cdot w' + \mathfrak{z} \cdot \omega + m \cdot \omega') d\xi,$$

$$W_a(w_a) = - \int_0^L H (z_a \cdot w_a + t_a \cdot w'_a) d\xi,$$

$$W_s(w_s) = - \int_0^L H (z_s \cdot w_s + t_s \cdot w'_s) d\xi,$$



$$H(\xi) = \begin{cases} 1 & \xi \in I_p \\ 0 & \xi \notin I_p \end{cases} \quad I_p := [l_p - L_p/2, l_p + L_p/2] \subset [0, L]$$

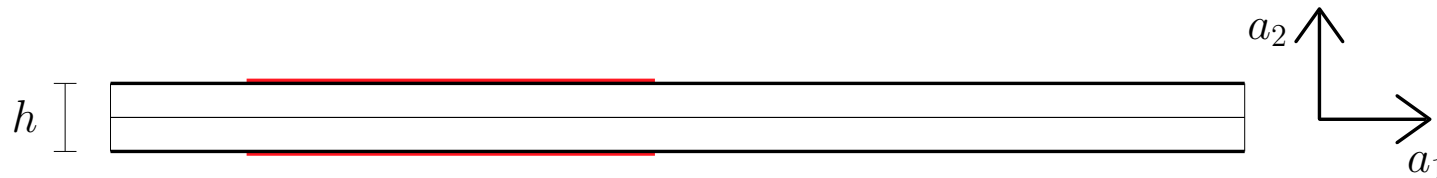
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$$W_s(w_s) = - \int_0^L H (z_s \cdot w_s + t_s \cdot w'_s) d\xi,$$



$$w_a := w + \omega \times \frac{h}{2} a_2, \quad w_s := w - \omega \times \frac{h}{2} a_2,$$

## Modello meccanico-elettrico

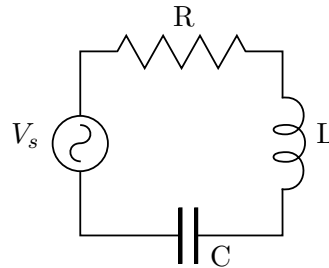
$$W_{out}(w, \omega) = \int_0^L (b \cdot w + c \cdot \omega) d\xi + \mathbf{t}_0 \cdot w(0) + \mathbf{t}_L \cdot w(L) + \mathbf{m}_0 \cdot \omega(0) + \mathbf{m}_L \cdot \omega(L),$$

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$$W_s(w_s) = - \int_0^L H (z_s \cdot w_s + t_s \cdot w'_s) d\xi,$$

$$W_c(I_s) = (L \ddot{Q}_s + R \dot{Q}_s + V_s) I_s.$$



## Obiettività materiale

$$W_b(w, \omega) = - \int_0^L (z \cdot w + t \cdot w' + \mathfrak{z} \cdot \omega + m \cdot \omega') d\xi$$

$$W_a(w_a) = - \int_0^L H (z_a \cdot w_a + t_a \cdot w'_a) d\xi$$

$$W_s(w_s) = - \int_0^L H (z_s \cdot w_s + t_s \cdot w'_s) d\xi$$

Per ogni atto di moto test rigido

$$\omega(\xi) = \bar{\omega}$$

$$w(\xi) = \bar{w} + \bar{\omega} \times (x(\xi) - x_0)$$

sia

$$z \cdot w + t \cdot w' + \mathfrak{z} \cdot \omega + m \cdot \omega' = 0$$

$$z_a \cdot w_a + t_a \cdot w'_a = 0$$

$$z_s \cdot w_s + t_s \cdot w'_s = 0$$

## Obiettività materiale

$$\begin{aligned}z &= 0, & \mathfrak{z} &= -x' \times t, \\z_a &= 0, & 0 &= x'_a \times t_a, \\z_s &= 0, & 0 &= x'_s \times t_s.\end{aligned}$$

$$W_b(w, \omega) = - \int_0^L (t \cdot w' - x' \times t \cdot \omega + m \cdot \omega') d\xi$$

$$W_a(w_a) = - \int_0^L (t_a H) \cdot w'_a d\xi$$

$$W_s(w_s) = - \int_0^L (t_s H) \cdot w'_s d\xi$$

## Equazioni di bilancio

$$W_{out} + W_b + W_a + W_s + W_c = 0, \quad \forall w, \omega \in C^1([0, L]), \forall I_s \in \mathbb{R}$$

Integrando per parti

$$\int_0^L (t_a H) \cdot w'_a \, d\xi = - \int_0^L (t_a H)' \cdot w_a \, d\xi$$

$$\int_0^L t \cdot w' \, d\xi = - \int_0^L t' \cdot w \, d\xi + t(L)w(L) - t(0)w(0)$$

si ottiene

$$\int_0^L (b + t' + (t_a H)' + (t_s H)') \cdot w \, d\xi$$

$$+ \int_0^L \left( c + x' \times t + m' + \frac{h}{2} a_2 \times (t_a H)' - \frac{h}{2} a_2 \times (t_s H)' \right) \cdot \omega \, d\xi$$

$$+ (\mathbf{t}_0 + t(0)) \cdot w(0) + (\mathbf{m}_0 + m(0)) \cdot \omega(0)$$

$$+ (\mathbf{t}_L - t(L)) \cdot w(L) + (\mathbf{m}_L - m(L)) \cdot \omega(L)$$

$$+ (L \ddot{Q}_s + R \dot{Q}_s + V_s) I_s = 0.$$



## Equazioni di bilancio

$$b + t' + (t_a H)' + (t_s H)' = 0$$

$$c + x' \times t + m' + \frac{h}{2} a_2 \times (t_a H)' - \frac{h}{2} a_2 \times (t_s H)' = 0$$

$$L \ddot{Q}_s + R \dot{Q}_s + V_s = 0$$

$$v(0) = 0, \quad t(L) = \mathbf{t}_L$$

$$\theta(0) = 0, \quad m(L) = \mathbf{m}_L$$

$$b := -\rho \ddot{x}$$

$$c := 0$$

$$\rho := \rho_b + (\rho_a + \rho_s) H$$

$$\mathbf{t}_L := -\mu \ddot{x}(L) + \mu r (\ddot{\theta}(L) a_1 + \dot{\theta}(L)^2 a_2)$$

$$\mathbf{m}_L := \mu r^2 \ddot{\theta}(L) a_3 - \mu r a_2 \times \ddot{x}(L)$$

$\mu$  massa dell'accelerometro

$r$  eccentricità dell'accelerometro

## Tensione interlaminare

$$b + t' + (t_a H)' + (t_s H)' = 0$$

$$x'_a \times t_a = 0$$

$$x'_s \times t_s = 0$$

$$(t_a(\xi) H(\xi))' = t'_a(\xi) H(\xi) + t_a(\xi) \delta(\xi - \ell_p + L_p/2) - t_a(\xi) \delta(\xi - \ell_p - L_p/2).$$

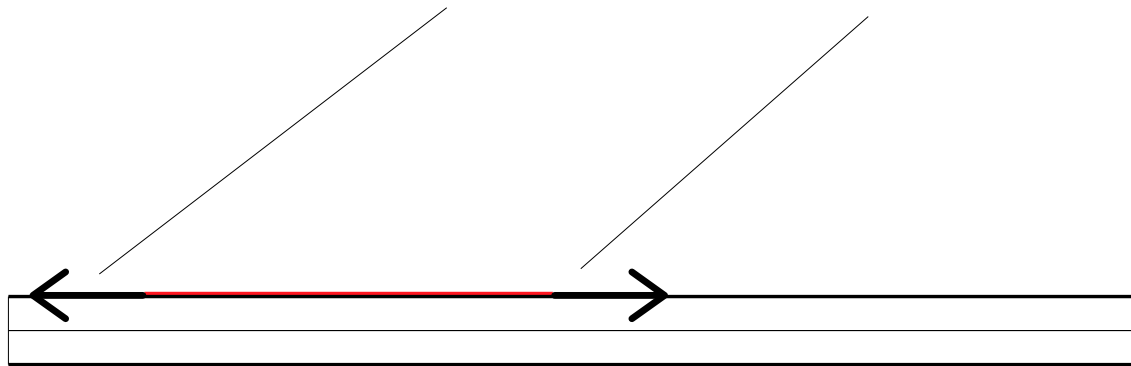
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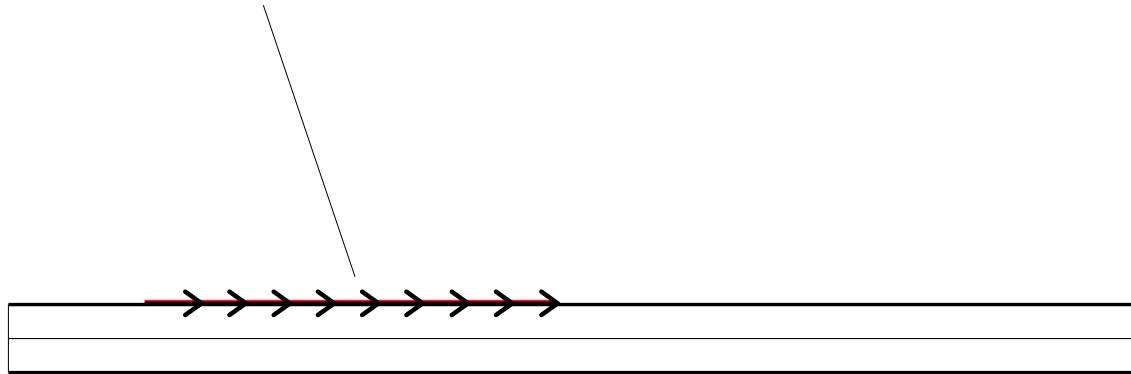
## Tensione interlaminare

$$b + t' + (t_a H)' + (t_s H)' = 0$$

$$x'_a \times t_a = 0$$

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$$(t_a(\xi) H(\xi))' = t'_a(\xi) H(\xi) + t_a(\xi) \delta(\xi - \ell_p + L_p/2) - t_a(\xi) \delta(\xi - \ell_p - L_p/2).$$



## Caratterizzazione costitutiva delle lamine piezoelettriche

$$t_s = N_s a_1$$

$$t_a = N_a a_1$$

attuatore

$$\begin{pmatrix} N_a \\ Q_a \end{pmatrix} = \begin{pmatrix} \bar{C}_{11}^E & \bar{e}_{31} \\ \bar{e}_{31} & -C \end{pmatrix} \begin{pmatrix} \Delta_a \\ V_a \end{pmatrix}$$

smorzatore

$$\begin{pmatrix} N_s \\ V_s \end{pmatrix} = \begin{pmatrix} \bar{C}_{11}^D & -\bar{\nu}_{31} \\ -\bar{\nu}_{31} & 1/C \end{pmatrix} \begin{pmatrix} \Delta_s \\ Q_s \end{pmatrix}$$

$$\Delta_s = -\Delta_a = \frac{h}{2} (\theta(\ell_p + L_p/2) - \theta(\ell_p - L_p/2))$$

$$\bar{C}_{11}^D = 1.28 \times 10^7 \text{ N/m},$$

$$\bar{C}_{11}^E = 8.62 \times 10^6 \text{ N/m},$$

$$\bar{\nu}_{31} = -3.77 \times 10^7 \text{ N/nF},$$

$$\bar{e}_{31} = -0.157 \text{ N/V},$$

$$C = 48.6 \text{ nF}.$$

## Equazioni lineari di bilancio

$$T' - \rho \ddot{v} = 0$$

$$M' + T - \frac{h}{2}(N_a H)' + \frac{h}{2}(N_s H)' = 0$$

$$v(0) = 0, \quad T(L) = \mu \ddot{v}(L),$$

$$\theta(0) = 0, \quad M(L) = \mu r^2 \ddot{\theta}(L),$$

## Equazioni lineari di bilancio

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$$\theta(0) = 0, \quad M(L) = \mu r^2 \ddot{\theta}(L),$$

$$M'' - \frac{h}{2}(N_a H)'' + \frac{h}{2}(N_s H)'' + \rho \ddot{v} = 0,$$

$$v(0) = 0, \quad -M'(L) = \mu \ddot{v}(L),$$

$$\theta(0) = 0, \quad M(L) = \mu r^2 \ddot{\theta}(L).$$

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$$\theta(0) = 0, \quad M(L) = \mu r^2 \ddot{\theta}(L).$$

$(N_a, N_s, \text{costanti})$

$$M'' - \frac{h}{2} N_a H'' + \frac{h}{2} N_s H'' + \rho \ddot{v} = 0$$



## Equazioni del moto

$$(\theta = v')$$

$$YJ v'''' + \frac{h^2}{4} (\bar{C}_{11}^E + \bar{C}_{11}^D) (v'(\ell_p + L_p/2) - v'(\ell_p - L_p/2)) H'' + \rho \ddot{v} = \frac{h}{2} \bar{\nu}_{31} Q_s H'' + \frac{h}{2} \bar{e}_{31} V_a H''$$

$$v(0) = 0, \quad -YJ v'''(L) = \mu \ddot{v}(L)$$

$$v'(0) = 0, \quad YJ v''(L) = \mu r^2 \dot{v}'(L)$$

$$L \ddot{Q}_s + R \dot{Q}_s + \frac{Q_s}{C} - \bar{\nu}_{31} \frac{h}{2} (v'(\ell_p + L_p/2) - v'(\ell_p - L_p/2)) = 0$$

## Equazioni modali

$$v(\xi, t) = \sum_{i=1}^N \Phi_i(\xi) X_i(t),$$

$$\ddot{X}_i + 2\omega_i \zeta_i \dot{X}_i + \omega_i^2 X_i + q_i Q_s = p_i V_a,$$

$$L \ddot{Q}_s + R \dot{Q}_s + \frac{Q_s}{C} + \sum_{i=1}^N q_i \|\Phi_i\| X_i = 0,$$

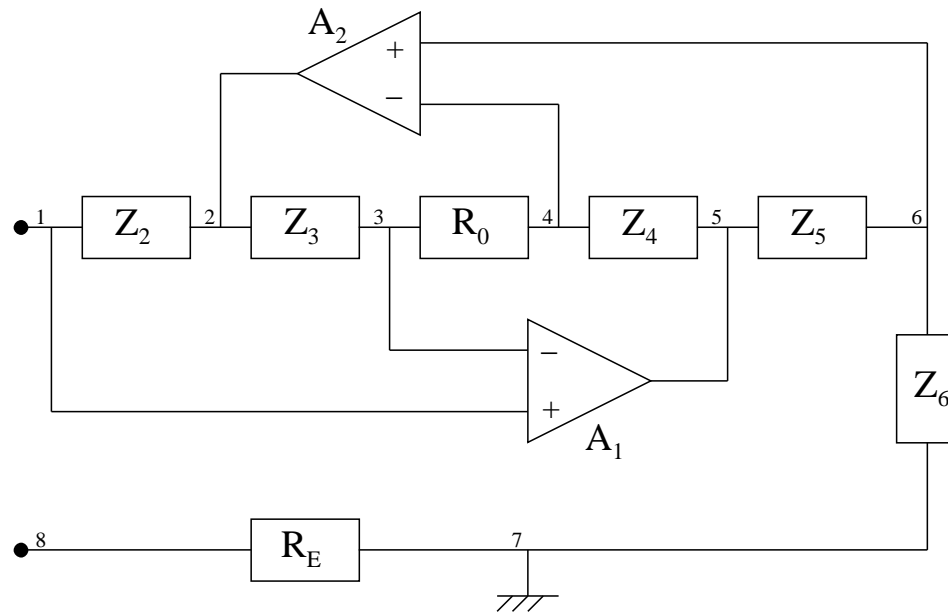
$$\|\Phi_i\| := \int_0^L \rho \Phi_i^2(\xi) d\xi + \mu \Phi_i(L)^2 + \mu r^2 \Phi_i'(L)^2,$$

$$q_i := -\frac{1}{\|\Phi_i\|} \int_0^L \frac{h}{2} \bar{\nu}_{31} \Phi_i(\xi) H''(\xi) d\xi = -\frac{h}{2} \bar{\nu}_{31} \frac{\Phi_i'(\ell_p + L_p/2) - \Phi_i'(\ell_p - L_p/2)}{\|\Phi_i\|},$$

$$p_i := \frac{1}{\|\Phi_i\|} \int_0^L \frac{h}{2} \bar{e}_{31} \Phi_i(\xi) H''(\xi) d\xi = \frac{h}{2} \bar{e}_{31} \frac{\Phi_i'(\ell_p + L_p/2) - \Phi_i'(\ell_p - L_p/2)}{\|\Phi_i\|}.$$

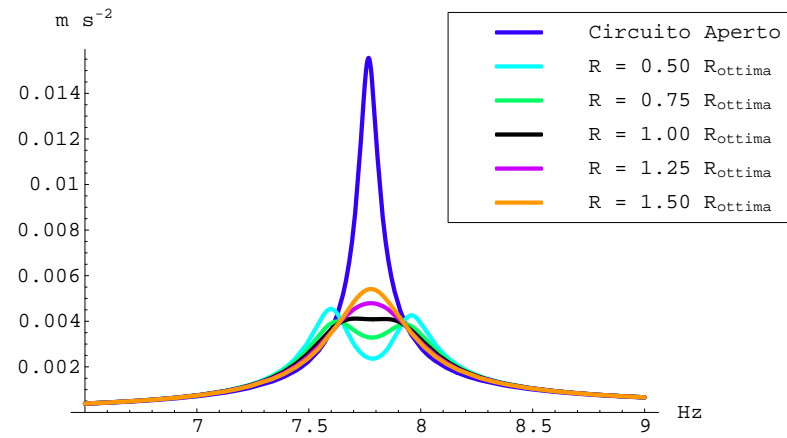
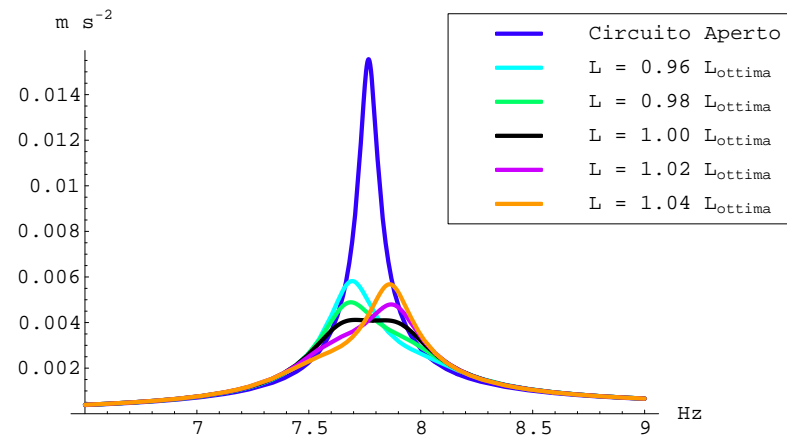
Barra di alluminio	Lamina Piezoelettrica
$L = 0.511 \text{ m}$	$L_p = 4.597 \times 10^{-2} \text{ m}$ $\ell_p = 8.4 \times 10^{-2} \text{ m}$
$h = 3.21 \times 10^{-3} \text{ m}$	$h_p = 2.54 \times 10^{-4} \text{ m}$
$J = 7.028 \times 10^{-11} \text{ m}^4$	$Y_{11}^E = 6.9 \times 10^{10} \text{ N/m}^2$
$Y = 67.6 \text{ GPa}$	$Y_{33}^E = 5.5 \times 10^{10} \text{ N/m}^2$
$\rho_b = 0.2236 \text{ Kg/m}$	$\rho_p = 2 \times 0.0616 \text{ Kg/m}$

# Giratore di Antoniou modificato

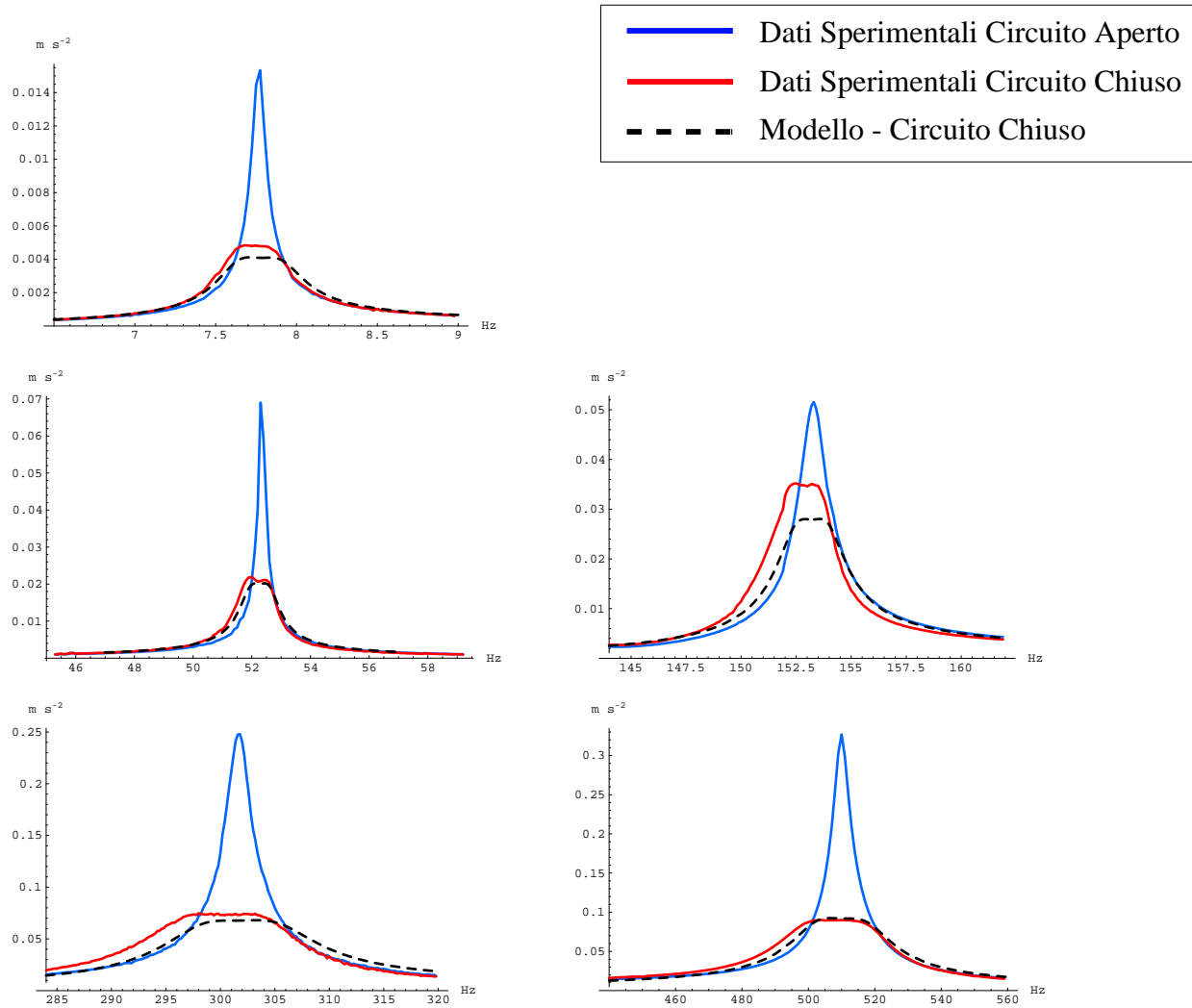


$$Z_{in} = \frac{Z_2 Z_4 Z_6}{Z_3 Z_5} - R_0 \frac{Z_2}{Z_3} + R_E$$

Giratore	1	2	3	4	5
frequenza (Hz)	7.775	52.3	153.3	301.8	510
$C$ (nF)	48.6	48.6	48.6	48.6	48.6
Operazionali $A_1$ - $A_2$	TL081 $\pm 15$ Volt				
$V_a$ (Volt)	0.7	0.1	0.1	0.01	0.05
$Z_2$ ( $K\Omega$ )	46.7	3	3	3	3
$Z_3$ ( $K\Omega$ ); $Z_4$ ( $K\Omega$ )	0.995	0.995	0.995	0.995	0.995
$Z_5$ ( $\mu F$ )	10	0.88	0.88	0.0221	0.0221
$Z_6$ ( $K\Omega$ )	17.52	70.6	8.52	93	33.4
$R_E$ ( $K\Omega$ )	3.88	0	0	0.0681	0.173
$R_0$ ( $K\Omega$ )	0	0.418	0.1961	0	0
$L = 1/(\omega^2 C)$ (Henry)	8622	190.6	22.18	5.722	2.004
$L_{ottima}$ (Henry)	<b>8562</b>	<b>190.1</b>	<b>22.13</b>	<b>5.694</b>	<b>2.002</b>
$L_{ottima}/L$	0.9930	0.9975	0.9979	0.9951	0.9993
$R_{ottima}$ ( $K\Omega$ )	<b>37.908</b>	<b>2.098</b>	<b>0.4657</b>	<b>0.6728</b>	<b>0.5073</b>
Attenuazione della risposta	69 %	70 %	32 %	71 %	73 %



# Confronto tra risposta calcolata e risposta misurata



## Conclusioni

- sensibile attenuazione dell'ampiezza delle oscillazioni
- buon accordo tra risposta calcolata e risposta misurata

## Problemi

- saturazione degli operazionali del giratore
- risposta del modello poco accurata per frequenze alte

## Proposti

- utilizzare nuova elettronica (giratore basato su *current-conveyors*)
- adottare nuovo modello costitutivo per le lamine piezoelettriche



## Riferimenti bibliografici

- [1] A. Fernandes, J. Pouget, E. Silverii, “Piezoelectric multilayer plates: sensor, actuator and vibration aspects”, in *Mechanics of Electromagnetic Solids*, Eds. J. Yang & G.A.M. Maugin, Kluwer Academic Publisher, Norwell, 2002.
- [2] E. Silverii, “Smorzamento attivo e passivo di vibrazioni strutturali con lamine piezoelettriche”, *Tesi di Dottorato in Ingegneria delle Strutture*, Roma-L’Aquila, 2002.
- [3] V. Gattulli, E. Silverii, A. Tatone, “Feedback Bending Control Experienced by Piezoelectric Actuators”, 3rd World Conference on Structural Control, 2002.
- [4] Seung Jo Kim, Chang Heon Han, Chul Yong Yun, “Improvement of aeroelastic stability of hingeless helicopter rotor blade by passive piezoelectric damping”, SPIE vol. 3672, 1999.
- [5] A. Tatone, E. Silverii and G. Rotoli, “Oscillazioni di una lamina metallica accoppiata ad una lamina piezoelettrica”, Congresso AIMETA’99, Como, 4-6 Ottobre 1999.
- [6] A. Tatone, “A one-dimensional model for multilayered planar beams”, *Proceedings SPIE’s 1996 Symposium on Smart Structures and Materials*, San Diego (CA), 1996.
- [7] Shu-yau Wu, “Piezoelectric shunts with a parallel R-L circuit for structural damping and vibration control”, *Proceedings of the International Society for Optical Engineering*,

SPIE vol. 2720, pp.259–269, 1996.

- [8] L. von Wangenheim, “Modification of the classical GIC structure and its application to RG oscillators”, *Electron. Lett.*, 32(1), pp. 6–8, 1996.
- [9] K.L. Tver, M.F. Golnaraghi and D. Wang, “Development of a generalized active vibration suppression strategy for cantilever beam using internal resonance”, *Nonlinear Dynamics*, 5(1), pp. 71–92, 1994.
- [10] N.W. Hagood, A. Von Flotow, ”Damping of structural vibrations with piezoelectric materials and passive electrical networks”, *Journal of Sound and Vibration*, 146(2), 243–268, 1991.
- [11] T. Ikeda, *Fundamentals of Piezoelectricity*, Oxford University Press, 1990.
- [12] T. Bailey, J.E. Hubbard , “Distributed Piezoelectric-Polymer Active Vibration Control of a Cantilever Beam”, *Journal of Guidance*, 8(5), pp. 605–611, 1985.
- [13] A. Antoniou, “Realisation of gyrators using operational amplifiers”, *IEE Proc.*, 116(11), pp. 1838–1850, 1969.